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APPLICATION NO.	FILING DATE	FIRST NAMED INVENTOR	ATTORNEY DOCKET NO.	CONFIRMATION NO.
10/757,419	01/15/2004	Berndt Brenner	P24717	3378
7055 7590 03/01/2007 GREENBLUM & BERNSTEIN, P.L.C. 1950 ROLAND CLARKE PLACE			EXAMINER	
			DINH, TIEN QUANG	
RESTON, VA 20191		•	ART UNIT	PAPER NUMBER
			3644	
SHORTENED STATUTOR	RY PERIOD OF RESPONSE	NOTIFICATION DATE	DELIVER	Y MODE
3 MO	NTHS	03/01/2007	ELECTRONIC	

Please find below and/or attached an Office communication concerning this application or proceeding.

If NO period for reply is specified above, the maximum statutory period will apply and will expire 6 MONTHS from the mailing date of this communication.

Notice of this Office communication was sent electronically on the above-indicated "Notification Date" and has a shortened statutory period for reply of 3 MONTHS from 03/01/2007.

Notice of the Office communication was sent electronically on above-indicated "Notification Date" to the following e-mail address(es):

gbpatent@gbpatent.com pto@gbpatent.com

	Application No.	Applicant(s)				
	10/757,419	BRENNER ET AL.				
Office Action Summary	Examiner	Art Unit				
	Tien Dinh	3644				
The MAILING DATE of this communication appears on the cover sheet with the correspondence address Period for Reply						
A SHORTENED STATUTORY PERIOD FOR REPL WHICHEVER IS LONGER, FROM THE MAILING ID. - Extensions of time may be available under the provisions of 37 CFR 1. after SIX (6) MONTHS from the mailing date of this communication. - If NO period for reply is specified above, the maximum statutory period. - Failure to reply within the set or extended period for reply will, by statut Any reply received by the Office later than three months after the mailing earned patent term adjustment. See 37 CFR 1.704(b).	DATE OF THIS COMMUNICATION 136(a). In no event, however, may a reply be to the state of the sta	DN. Imply filed Imply filed				
Status						
1)⊠ Responsive to communication(s) filed on <u>30 November 2006</u> .						
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closed in accordance with the practice under Ex parte Quayle, 1935 C.D. 11, 453 O.G. 213.						
Disposition of Claims						
4)⊠ Claim(s) <u>1-8,11-66,69-104 and 136-145</u> is/are pending in the application.						
4a) Of the above claim(s) <u>See Continuation Sheet</u> is/are withdrawn from consideration.						
5) Claim(s) <u>12,13,16,17 and 21</u> is/are allowed.						
6)⊠ Claim(s) <u>1-5,7,8,14,15,20,24-27,49,50,55,58-63,65,66,69-76,97,136,137 and 140-145</u> is/are rejected.						
7)⊠ Claim(s) <u>11</u> is/are objected to.						
8) Claim(s) are subject to restriction and/	or election requirement.					
Application Papers						
9) The specification is objected to by the Examiner.						
10) The drawing(s) filed on is/are: a) accepted or b) objected to by the Examiner.						
Applicant may not request that any objection to the drawing(s) be held in abeyance. See 37 CFR 1.85(a).						
Replacement drawing sheet(s) including the correction is required if the drawing(s) is objected to. See 37 CFR 1.121(d).						
11) The oath or declaration is objected to by the Examiner. Note the attached Office Action or form PTO-152.						
Priority under 35 U.S.C. § 119						
12) Acknowledgment is made of a claim for foreign priority under 35 U.S.C. § 119(a)-(d) or (f). a) All b) Some * c) None of:						
 Certified copies of the priority documents have been received. 						
2. Certified copies of the priority documents have been received in Application No						
3. Copies of the certified copies of the priority documents have been received in this National Stage						
application from the International Bureau (PCT Rule 17.2(a)).						
* See the attached detailed Office action for a list of the certified copies not received.						
Attachment(s)						
 Notice of References Cited (PTO-892) Notice of Draftsperson's Patent Drawing Review (PTO-948) 	4) Interview Summa Paper No(s)/Mail					
 2) Notice of Draftsperson's Patent Drawing Review (PTO-948) 3) Information Disclosure Statement(s) (PTO/SB/08) Paper No(s)/Mail Date 		I Patent Application				

Continuation of Disposition of Claims: Claims withdrawn from consideration are 6,18,19,22,23,28-48,51-54,56,57,64,77-96,98-104,138 and 139.

DETAILED ACTION

Election/Restrictions

In response to applicant's arguments that the examiner improperly withdrew claims 6, 18, 19, 22, 23, 28-48, 51-54, 56, 57, 64, 77-96, 98-104, 138, and 139, the examiner would like to point out that there were two restriction/election requirements that were sent out. The first was on 6/6/05 and the second was 2/15/06. The first election in response of the office action was to elect group A as indicated in the response on 12/06/05. The second election was 3/17/06 in response to the restriction sent out on 2/15/06. The claims were properly withdrawn.

Claim Rejections - 35 USC § 103

The following is a quotation of 35 U.S.C. 103(a) which forms the basis for all obviousness rejections set forth in this Office action:

(a) A patent may not be obtained though the invention is not identically disclosed or described as set forth in section 102 of this title, if the differences between the subject matter sought to be patented and the prior art are such that the subject matter as a whole would have been obvious at the time the invention was made to a person having ordinary skill in the art to which said subject matter pertains. Patentability shall not be negatived by the manner in which the invention was made.

Claims 1-5, 7-8, 14, 15, 20, 24-27, 49, 50, 55, 58, 62, 63, 65, 66, 69, 97, 136, 137, 140-145 are rejected under 35 U.S.C. 103(a) as being unpatentable over Mueller et al in view of Palm. Bauer, and Schmidt.

Mueller et al teaches a lightweight structural component (that can be oriented lengthwise and crosswise) having panel/skin sheet 40, side pieces made up of parts 26, 27, 40 that form an angle of 7-50 degrees, and stringer made up of parts 21-25. Please note that the head portion is element 27' that is coupled to a bar portion in figure 1. Mueller et al is silent on the thickened region/stiffening base in the panel, the metal to make up the panel and the stiffening elements,

and the use of "weld joints". However, Schmidt discloses that a thickened region in a panel 4 where a side piece 7a is connected to is well known. Schmidt discloses that using welding techniques such as laser beam welding and other know techniques such as friction stir, etc. to combine two parts are well known. See column 5, lines 51-60. Palm discloses that a support element 2 that is metal that is attached to a metal panel 1 via welding is well known in the art. Please note that both Palm and Schmidt disclose that side pieces are welded to the thickened area of the panel. As further support, the examiner has cited Bauer to teach that use of metal or composites are merely substitution of parts (see column 2, lines 44-47).

It would have been obvious to one skilled in the art at the time the invention was made to have made the panel and the stiffener of Mueller et al out of metal and to use a thickened region on the panel as taught by Schmidt, Bauer, and Palm to increase make the aircraft structure to be stronger, easier to build, cheaper, and to prevent cracks.

Re claims 2 and 63, please note that the stiffener element is oriented lengthwise and crosswise since its base parts that are connected to the panel runs lengthwise and the stiffener run crosswise since it is running into and out of the page as viewed from figure 1.

Re claim 5, stringers are defined as "a longitudinal reinforcement in the fuselage or wing of an airplane". Mueller et al's stiffeners are stringers.

Re claims 20 and 145, please note that to make the components of the component integral are steps that is routine to one skilled in the art would have made to make a stronger structure.

See Nerwin v. Erlichman, 168 USPQ 177, 179. Please also note that stiffeners that are integral are disclosed Palm. One skilled in the art would have made stiffeners integral to make them stronger.

RE claim 24, please note that Mueller et al discloses edge areas 27', 27 that are parallel to the panel.

RE claim 27, please note that Mueller et al as modified by Palm and Schmidt disclose first and second base portions. This would mean that once a first and second base portions are put into Mueller et al's system, the base portions rest adjacent to the inner surfaces of the two side pieces.

RE claims 49-50, please note that the use of metal cutting, metal removal, deformed during rolling-in, etc. as claimed are obvious parts/steps that one skilled in the art would have used to make a stronger aircraft. The applicant has not cited the criticality to these claimed parts either.

Re claim 136, please note that the use of metal on a panel and stiffening element is a well known idea in this day and age to make the structure stronger. See prior arts noted above.

Please also note that the bar portion first thickness is where element 121 is located. This thickness is greater than the side pieces' thickness.

Re claim 140, the distance between the two separate joint zones is greater than the first thickness. See figure 1.

Re claim 141, the distance between the two separate joint zones is greater than the second thickness. See figure 1.

Re claim 142, the distance between the two inner edges of the two separate joint zones is greater than the first thickness. See figure 1.

Re claim 143, the distance between the two inner edges of the two separate joint zones is greater than the second thickness. See figure 1.

RE claim 144, Mueller et al as modified by Palm, Bauer, and Schmidt disclose two weld joint zones that are arranged between the thickened region of the panel and the two side pieces. Since the stiffeners are mounted on the thickened region of the panel and are welded there, this meets the limitation of the claim.

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Claims 59-61 and 70-76 are rejected under 35 U.S.C. 103(a) as being unpatentable over Mueller et al as modified by Schmidt in view of Lackman et al.

Mueller et al as modified by Schmidt discloses all claimed parts (including rough profile) except for reinforcing element. However, Lackman et al teaches that reinforcing element (in shape of an isosceles triangle is well known in the art.

It would have been obvious to one skilled in the art to have used reinforcing element in Mueller et al's system as modified by Schmidt and as taught by the Lackman et al to make the structure even stronger.

Re claim 60, it would have been obvious to one skilled in the art to have used a material of higher modulus of elasticity in the reinforcing element than those of the panel and stiffening element so that the aircraft can absorb more vibration and be made stronger. This merely involves a routine skill to one skilled in the art.

RE claim 61, the use of force-locking and form-locking manner is well known steps that one skilled in the art would have taken to make the structure stronger.

Re claim 70, the reinforcing element are forced locked and form locked to the inner surfaces of the side pieces and the thickened region.

Re claim 71, the surface is surface profiling.

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Re claim 75-76, please note that since Mueller teaches a rounded off apex in the cavity and Lackman et al teaches reinforcing element that fits snuggly to the cavity, once skilled in the art would have used reinforcing elements that have rounded off apex to accommodate the cavity of Mueller.

Response to Arguments

The examiner has used new prior arts to reject the amended claims. This renders some of applicant's arguments moot.

In response to applicant's argument that there is no suggestion to combine the references, the examiner recognizes that obviousness can only be established by combining or modifying the teachings of the prior art to produce the claimed invention where there is some teaching, suggestion, or motivation to do so found either in the references themselves or in the knowledge generally available to one of ordinary skill in the art. See *In re Fine*, 837 F.2d 1071, 5 USPQ2d 1596 (Fed. Cir. 1988)and *In re Jones*, 958 F.2d 347, 21 USPQ2d 1941 (Fed. Cir. 1992). In this case, Mueller et al in view of Palm, Bauer, and Schmidt discloses that an integral stiffener being made out of metal that are welded at two joint zones to have a stronger aircraft structure.

RE claims 4, 7-10, 20, 26, 27, 49, 50, 62, 63, 65-69, 97, 137, 140-143, the applicant does not seem to argue the details of such claims but seem to say that if the independent claims are allowable, then these dependent claims are allowable. The examiner disagrees. These claims are rejected with the new prior arts.

Re arguments about Lackman, the examiner fails to understand applicant's arguments on why since Lackman discloses a composite material instead of a metal material that it is not combinable with Mueller et al in view of Palm, Bauer, and Schmidt. The claims called for the reinforcing element to have modulus of elasticity. This means that it is a composite does it not? Please also note that the reinforcing element in Mueller et al's system would make the stiffener stronger. This is a valid motivation to combine.

RE claims 59-61 and 70-76, the applicant does not seem to argue the details of such claims but seem to say that if the independent claims are allowable, then these dependent claims are allowable. The examiner disagrees. Mueller et al in view of Palm, Bauer, and Schmidt and Lackhman disclose what is claimed in claims 59-61 and 70-76. This was presented in the last office action.

The examiner would like to point out that if the limitation of claim 11 were to be put into the independent claims, this application is most likely be in condition for allowance.

Allowable Subject Matter

Claims 12, 13, 16, 17, and 21 are allowed.

Claim 11 is objected to as being dependent upon a rejected base claim, but would be allowable if rewritten in independent form including all of the limitations of the base claim and any intervening claims.

Conclusion

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Applicant's amendment necessitated the new ground(s) of rejection presented in this Office action. Accordingly, **THIS ACTION IS MADE FINAL**. See MPEP § 706.07(a). Applicant is reminded of the extension of time policy as set forth in 37 CFR 1.136(a).

A shortened statutory period for reply to this final action is set to expire THREE MONTHS from the mailing date of this action. In the event a first reply is filed within TWO MONTHS of the mailing date of this final action and the advisory action is not mailed until after the end of the THREE-MONTH shortened statutory period, then the shortened statutory period will expire on the date the advisory action is mailed, and any extension fee pursuant to 37 CFR 1.136(a) will be calculated from the mailing date of the advisory action. In no event, however, will the statutory period for reply expire later than SIX MONTHS from the date of this final action.

Any inquiry concerning this communication or earlier communications from the examiner should be directed to Tien Dinh whose telephone number is 571-272-6899. The examiner can normally be reached on 9-6.

If attempts to reach the examiner by telephone are unsuccessful, the examiner's supervisor, Teri Luu can be reached on 571-272-7045. The fax phone number for the organization where this application or proceeding is assigned is 571-273-8300.

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